

EN 1993-1-1 **article 5.3.2(11)**:

As an alternative the shape of the elastic buckling mode  $\eta_{cr}$  of the structure may be applied as a unique global and local imperfection. The amplitude  $h_{init}$  of this imperfection may be determined from:

$\alpha$  - The imperfection factor for the relevant buckling curve.

$\chi$  - The reduction factor for the relevant buckling curve, depending on the relevant cross-section.

$N_{Rk}$  - The characteristic resistance to normal force of the critical cross-section, i.e.  $N_{pl,Rk}$ .

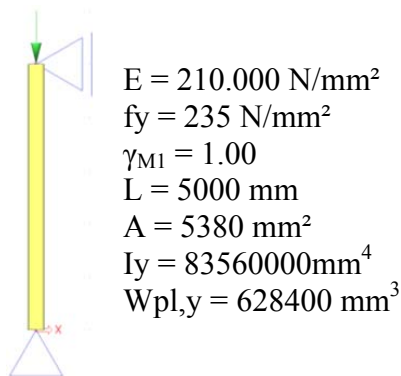
$N_{cr}$  - Elastic critical buckling load.

$M_{Rk}$  - The characteristic moment resistance of the critical cross-section, i.e.  $M_{el,Rk}$  or  $M_{pl,Rk}$  as relevant.

$\eta_{cr}$  - Shape of the elastic critical buckling mode.

$\eta''_{cr}$  - Maximal second derivative of the elastic critical buckling mode.

The column has a cross-section of type **IPE 300**, is fabricated from **S235** and has the following relevant properties:



First a **Stability calculation** is done using a load of 1kN. This way, the elastic critical buckling load  $N_{cr}$  is obtained. In order to obtain precise results, the **Number of 1D elements** is set to **10**. In addition, the **Shear Force Deformation** is neglected so the result can be checked by a manual calculation.

The stability calculation gives the following result:

### Critical load coefficients

Critical load coefficients	
N	f
-	∞
<b>Stability combination : S1</b>	
1	6885,28

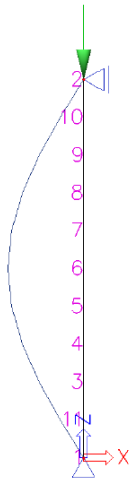
This can be verified with Euler's formula using the member length as the buckling length:

$$N_{cr} = \frac{\pi^2 EI}{l^2} = \frac{\pi^2 \times 210.000 \text{ N/mm}^2 \times 83560000 \text{ mm}^4}{(5000 \text{ mm})^2} = 6927.5 \text{ kN}$$

The following picture shows the mesh nodes of the column and the corresponding buckling shape:

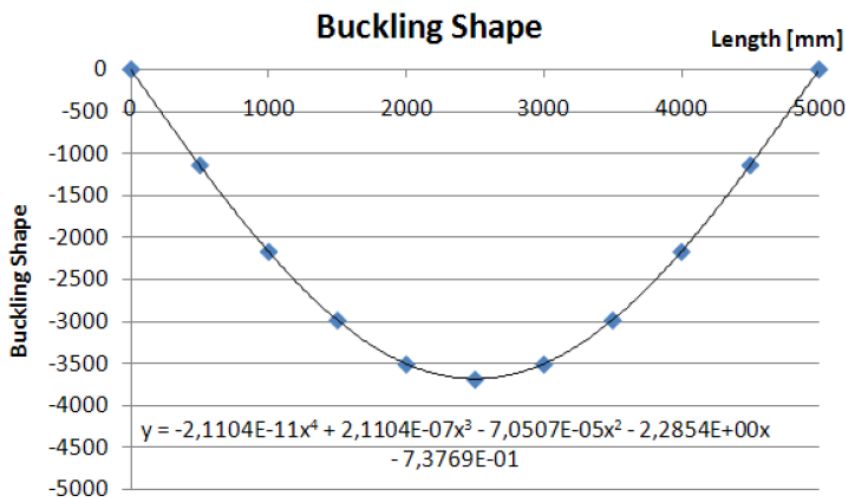
### Displacement of nodes

Stability calculation, Extreme : No  
 Selection : All  
 Stability combinations : S1/1 - 6885,28  
 Values are multiplied by 10000 for better numeric representation.



Node of mesh	Case	Ux [-]	Uz [-]	Fiy [-]
1	S1/1 - 6885,28	0,00	0,00	-2314,62
11	S1/1 - 6885,28	-1138,38	0,00	-2201,33
3	S1/1 - 6885,28	-2165,34	0,00	-1872,57
4	S1/1 - 6885,28	-2980,33	0,00	-1360,50
5	S1/1 - 6885,28	-3503,59	0,00	-715,26
6	S1/1 - 6885,28	-3683,89	0,00	0,00
7	S1/1 - 6885,28	-3503,59	0,00	715,26
8	S1/1 - 6885,28	-2980,33	0,00	1360,50
9	S1/1 - 6885,28	-2165,34	0,00	1872,57
10	S1/1 - 6885,28	-1138,38	0,00	2201,33
2	S1/1 - 6885,28	0,00	0,00	2314,62

Using for example an Excel worksheet, the buckling shape can be approximated by a 4th grade polynomial.



A polynomial has the advantage that the second derivative can easily be calculated.

$$\eta_{cr} = -2.1104 \times 10^{-11} \times x^4 + 2.1104 \times 10^{-7} \times x^3 - 7.0507 \times 10^{-5} \times x^2 + 2.2854 \times x - 7.3769 \times 10^{-1}$$

$$\eta''_{cr} = -2.5325 \times 10^{-10} \times x^2 + 1.2662 \times 10^{-6} \times x - 1.4101 \times 10^{-4}$$

Calculation of  $e_0$

$$N_{Rk} = f_y \times A = 235 \text{ N/mm}^2 \times 5380 \text{ mm}^2 = 1264300 \text{ N}$$

$$M_{Rk} = f_y \times W_{pl} = 235 \text{ N/mm}^2 \times 628400 \text{ mm}^3 = 147674000 \text{ Nmm (class 2)}$$

$$\bar{\lambda} = \sqrt{\frac{N_{Rk}}{N_{cr}}} = \sqrt{\frac{1264300 \text{ N}}{6927510 \text{ N}}} = 0.43$$

$\alpha = 0.21$  for buckling curve a

$$\chi = \frac{1}{0.5 [1 + \alpha(\bar{\lambda} - 0.2) + (\bar{\lambda})^2] + \sqrt{(0.5 [1 + \alpha(\bar{\lambda} - 0.2) + (\bar{\lambda})^2])^2 - (\bar{\lambda})^2}} = 0.945$$

$$e_0 = \alpha(\bar{\lambda} - 0.2) \frac{M_{Rk}}{N_{Rk}} \frac{1 - \frac{\chi \times (\bar{\lambda})^2}{\gamma_{M1}}}{1 - \chi \times (\bar{\lambda})^2} = 0.21 \times (0.43 - 0.2) \frac{147674000 \text{ Nmm}}{1264300 \text{ N}} = 5.6416 \text{ mm}$$

Calculation of  $\eta_{init}$

The mid section of the column is decisive  $\Rightarrow x = 2500 \text{ mm}$

$\eta_{cr}$  at mid section = -3681,8

$\eta''_{cr}$  at mid section =  $1,4418 \times 10^{-3} \text{ 1/mm}^2$

$$\eta_{init} = e_0 \times \frac{N_{cr}}{E \times I \times \eta''_{cr}} \times \eta_{cr} = 5.6416 \text{ mm} \times \frac{6927510 \text{ N}}{210000 \text{ N/mm}^2 \times 83560000 \text{ mm}^4 \times 1.4418 \times 10^{-3} \text{ 1/mm}^2} \times$$

$$3681.8 = 5.6528 \text{ mm}$$

This value can now be inputted as amplitude of the buckling shape for imperfection.